

Problem 3.1 - Übungsblatt 3

Two particles with masses $m_1 = m$ and $m_2 = 3m$ are moving in 3D space and interact through the potential

$$U(r) = \begin{cases} -V, & r \leq R \\ 0, & \text{otherwise} \end{cases} \quad V, R > 0$$

The energy E and the angular momentum l are given in the reference system of the relative coordinate, r .

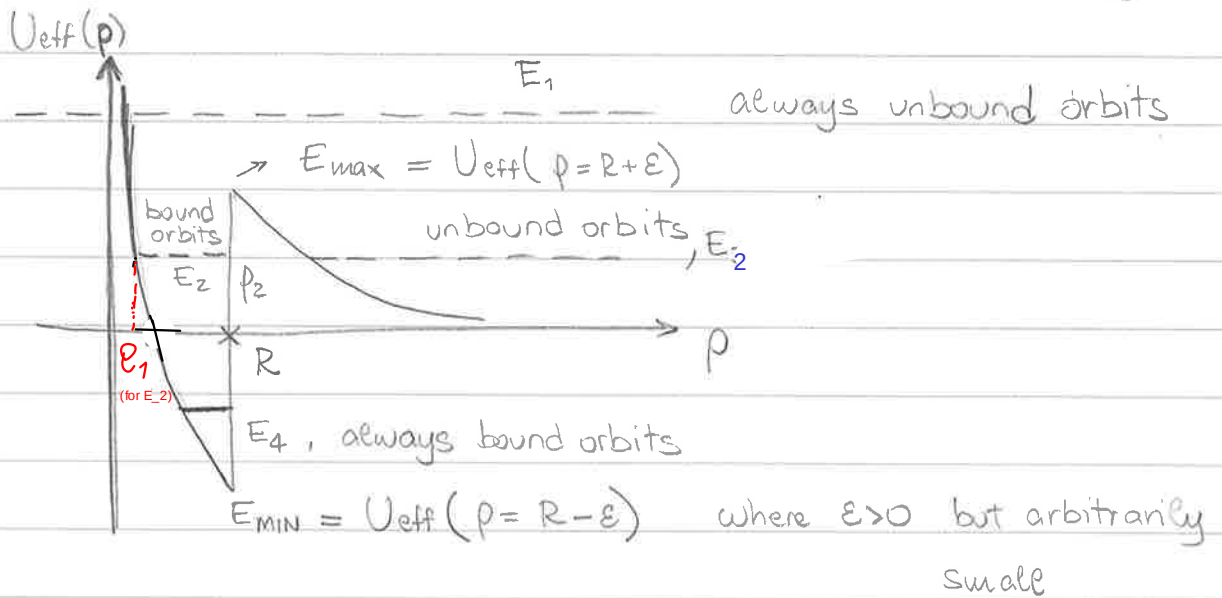
- 1) For which values of E are there **bounded** orbits?
- 2) Are there values of E for which both **bounded** and **unbound** orbits are possible?
- 3) Compute the minimum (p_1) and maximum (p_2) distances between the particles and only p_2 are open.
- 4) Determine the angle $\Delta\varphi$ by which the orbit rotates between two points in time at which the particles are at the maximum distance from each other. In which case does the orbit close: always, never, sometimes?
- 5) Assuming that a particle is at rest at the beginning and that the other is at a distance $R/2$ from the first and is moving with velocity v perpendicular to the

connecting axis, at which values of n are ^{there} bound orbits? Values of E and p are to be determined now.

Solution

Reference frame of the relative coordinate ρ

\Rightarrow center of mass moves with constant velocity \vec{v}



We work in the relative coordinates, there we have to use the reduced mass μ

We compute the reduced mass μ , i.e.

$$\mu \stackrel{\text{def}}{=} \frac{m_1 m_2}{m_1 + m_2} = \frac{3m^2}{4m} = \frac{3}{4}m$$

$m_1 = m$
 $m_2 = 3m$

$$U(\rho) = \begin{cases} -V & \rho \leq R \\ 0 & \text{otherwise} \end{cases} = -V \theta(R - \rho)$$

$$U_{\text{eff}}(\rho) = \frac{\ell^2}{2\mu\rho^2} - V \quad (\text{if } \rho \leq R) \quad U_{\text{eff}}(\rho) = \frac{\ell^2}{2\mu\rho^2}, \rho > R$$

With this we can now compute the inversion point, i.e.

$$E - U_{\text{eff}}(\rho) = 0$$

For $p \leq R$ (*)

$$E + V - \frac{l^2}{2\mu p^2} = 0$$

$$\frac{2\mu(E+V)}{l^2} = \frac{1}{p^2} \Rightarrow p_1 = \frac{l}{[2\mu(E+V)]^{1/2}}$$

So we the condition (*) can be further

restricted to being $p_1 = \frac{l}{\sqrt{2\mu(E+V)}} < R$

$$\Rightarrow E > \underbrace{\frac{l^2}{2\mu R^2} - V}_{E_{\min} = U_{\text{eff}}(p=R-E)}$$

$$E_{\min} = U_{\text{eff}}(p=R-E)$$

$\forall \epsilon > 0$ (arbitrarily small)

but still it has to be $E < E_{\max} = \frac{l^2}{2\mu R^2} = U_{\text{eff}}(R+E)$

So for bound orbits it must be

$$\boxed{\frac{l^2}{2\mu R^2} - V < E < \frac{l^2}{2\mu R^2}}$$

$$E_{\min} < E < E_{\max}$$

FOR $0 < E < E_{\max}$ THERE CAN BE BOTH BOUND AS WELL AS UNBOUND ORBITS DEPENDING ON INITIAL CONDITIONS
FOR $E > E_{\max}$ ONLY UNBOUND ORBITS

Computing the angle $\Delta\varphi$:

$$\Delta\varphi = 2 \int_{p_1}^R \frac{dp}{\sqrt{\tilde{F}(p)}}$$

Where we use the formula:

$$\begin{aligned}\tilde{F}(p) &= \frac{\mu^2 p^4}{e^2} \left[\frac{2}{M} \left(E+V - \frac{e^2}{2\mu p^2} \right) \right] = \\ &= \frac{2\mu p^4}{e^2} (E+V) - p^2\end{aligned}$$

For $p < R$ we have

$$\tilde{F}(p) = C^2 p^4 - p^2 \quad \text{with} \quad C^2 := \frac{2\mu(E+V)}{e^2} \equiv \frac{1}{p_1^2}$$

$$\Delta\varphi = 2 \int_{p_1}^R dp \frac{1}{\sqrt{\frac{p^4}{p_1^2} - p^2}} =$$

from our previous calculations

$$= 2 \int_{p_1}^R \frac{dp}{p} \frac{1}{\sqrt{\left(\frac{p}{p_1}\right)^2 - 1}} \quad \left(p = p_1 x, \quad dp = p_1 dx \right)$$
$$= 2 \int_1^{R/p_1} \frac{dx}{x \sqrt{x^2 - 1}}$$

$$= 2 \cos^{-1} \left(\frac{p_1}{R} \right)$$

$$\Delta\varphi = 2 \cos^{-1} (p_1/R)$$

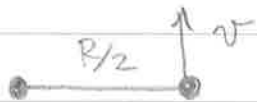
In order for the orbit to close $\Delta\varphi$ must be some rational number times 2π

$$\text{i.e. } \Delta\varphi = 2\pi \cdot \frac{n}{m}$$

FOR EXAMPLE FOR $p_1 = \frac{R}{2} \Rightarrow \Delta\varphi = \frac{\pi}{6}$ CLOSES

ALSO $p_1 = \frac{1}{\sqrt{2}} R \Rightarrow \Delta\varphi = \frac{\pi}{4}$ CLOSES.

IN GENERAL IT DOES NOT CLOSE



particle 1
(at rest)

particle 2
(moving)

The energy in the relative coordinates is given by

$$E_r = \frac{\mu v^2}{2} - V \quad (\text{since } p < R) \\ (\text{they're closer than } R)$$

the angular momentum

$$l_r = \mu \frac{R v}{2}$$

$$\frac{l_r^2}{2\mu R^2} = \frac{\mu^2 R^2 v^2}{4} \frac{1}{2\mu R^2} = \frac{\mu v^2}{8}$$

The condition for bound orbits then reads

$$\left(E_{\min} = \frac{l_r^2}{2\mu R^2} - V < E_r < \frac{l_r^2}{2\mu R^2} = E_{\max} \right)$$

$$\Rightarrow \frac{\mu v^2}{8} - V < E_r < \frac{\mu v^2}{8}$$

$$\uparrow \\ E_r + V = \frac{\mu v^2}{2}$$

$$\frac{\mu v^2}{8} - V < \frac{\mu v^2}{2} - V < \frac{\mu v^2}{8}$$

$$\frac{\mu v^2}{8} < \underbrace{\frac{\mu v^2}{2}} < \frac{\mu v^2}{8} + V$$

$$\frac{\mu v^2}{2} < \frac{\mu v^2}{8} + V$$

$$\frac{\mu v^2}{2} \left(1 - \frac{1}{4}\right) < V$$

$$\frac{3\mu v^2}{8} < V \Rightarrow \frac{3}{8} \frac{3m}{4} v^2 < V \Rightarrow \frac{9}{16} m v^2 < V$$

\uparrow
 $\mu = \frac{3}{4}m$

$$v < \frac{4}{3} \sqrt{\frac{2V}{m}}$$

Condition for bound orbits

3.2

$$\vec{r}_1 = (d, 0, 0)$$

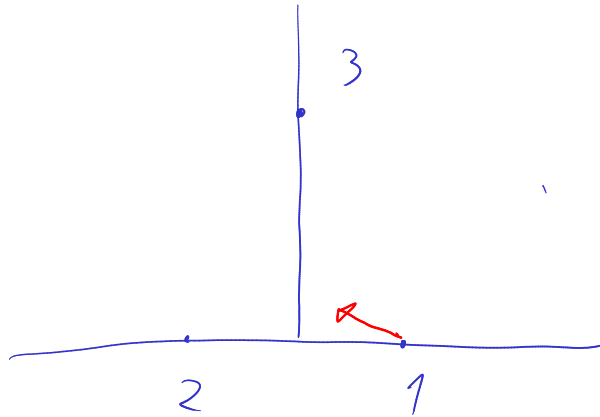
$$m_1 = m$$

$$\vec{r}_2 = (-d, 0, 0)$$

$$m_2 = m$$

$$\vec{r}_3 = (0, 2d, 0)$$

$$m_3 = 2m$$



KRAFT VON i AUF j

$$\vec{F}_{ij} = \alpha |\vec{r}_{ij}|^2 \vec{r}_{ij}$$

$$\vec{r}_{ij} = \vec{r}_i - \vec{r}_j$$

Zeigt in Richtung i : attraktiv

$\propto \vec{r}_{ij} \rightarrow$ zentral

$\propto \vec{e}_{ij} f(|\vec{r}_{ij}|) \rightarrow$ ISOTROP

$$\vec{r}_{12} = 2d \vec{e}_x$$

$$\vec{r}_{13} = d \vec{e}_x - 2d \vec{e}_y$$

$$\vec{r}_{23} = -d \vec{e}_x - 2d \vec{e}_y$$

$$\vec{F}_1 = -\alpha d^3 (8 \vec{e}_x + 5 \cdot (e_x - 2e_y))$$

$$= -\alpha d^3 (13 \vec{e}_x - 10 \vec{e}_y)$$

Potentialenergie Zuerst berechne die Potentialenergie U_{ij}
zwischen Teilchen i und j

es gilt $\vec{F}_{ij} = - \vec{\nabla}_{\vec{r}_j} U(\vec{r}_j - \vec{r}_i)$

setze $\vec{r} = \vec{r}_j - \vec{r}_i \Rightarrow \vec{\nabla}_{\vec{r}_j} = \vec{\nabla}_{\vec{r}}$
kettenregel

$$\vec{F}_{ij} = - d r^2 \vec{r} = - \nabla U(r)$$

\uparrow weil im Ausdruck steht $\vec{r}_i - \vec{r}_j = -\vec{r}$

$$= - d r^3 \frac{\vec{r}}{r} = - d r^3 \vec{e}_r$$

$$F = f(r) \vec{e}_r$$

$\rightarrow U(r)$

$$-\nabla U(r) = - \frac{\partial U}{\partial r} \vec{e}_r \Rightarrow U(r) = \frac{d r^4}{4} + U_0$$

$U_0 = 0$ weil $U=0$ wenn $r=0$

$$U_{GES} = U(\vec{r}_1 - \vec{r}_2) + U(\vec{r}_1 - \vec{r}_3) + U(\vec{r}_2 - \vec{r}_3) = \frac{d}{4} d^4 (4^2 + 5^2 + 5^2) = \frac{d}{4} d^4 \cdot 66$$
$$= \frac{33}{2} d^4$$

3.3

$$\dot{\vec{r}}_1 = u \vec{e}_y = \dot{\vec{r}}_2$$

$$\dot{\vec{r}}_3 = -u \vec{e}_y + 2u \vec{e}_x$$

$$\vec{L}_1 = m d u \vec{e}_x \times \vec{e}_y = m d u \vec{e}_z$$

$$\vec{L}_2 = -m d u \vec{e}_z$$

$$\vec{L}_3 = 2m d u (2 \vec{e}_y) \times (-\vec{e}_y + 2 \vec{e}_x) = -8 m d u \vec{e}_z$$

$$\vec{L} = -8 m d u \vec{e}_z$$

$$T = \frac{1}{2} m u^2 (1+1) + \frac{1}{2} 2m u^2 (5)$$

$$= m u^2 (1+5) = 6 m u^2$$

$$U = \frac{33}{2} 2 d^4$$

$$E = T + U$$

(3.4)

SCHWERPUNKTSYSTEM

$$\vec{R} = \frac{d}{4m} m (\vec{l}_x - \vec{l}_x + 2 \cdot 2 \vec{l}_y) = d \vec{l}_y$$

$$\vec{r}_1^{BS} = d (\vec{l}_x - \vec{l}_y)$$

$$\vec{r}_2^{BS} = d (-\vec{l}_x - \vec{l}_y)$$

$$\vec{r}_3^{BS} = d \vec{l}_y$$

$$\dot{\vec{R}} = \frac{\mu}{4m} m (\dot{\vec{l}}_y + \dot{\vec{l}}_y - 2 \dot{\vec{l}}_y + 4 \dot{\vec{l}}_x) = \mu \dot{\vec{l}}_x$$

$$\dot{\vec{r}}_1^{BS} = \mu (\dot{\vec{l}}_y - \dot{\vec{l}}_x) = \dot{\vec{r}}_2^S$$

$$\dot{\vec{r}}_3^{BS} = \mu (\dot{\vec{l}}_x - \dot{\vec{l}}_y)$$

$$T^{BS} = \frac{1}{2} m \mu^2 (2 + 2 + 2 \cdot 2) = 4m\mu^2$$

$$T_2 = \frac{1}{2} 4m \mu^2 \cdot 1 = 2m\mu^2$$

$$T = T^{BS} + T_2 = 6m\mu^2 \quad \checkmark$$

$$U = U^{BS}$$

DA ABSTÄNDE GLEICH BLEIBEN

$$\Rightarrow E = \underbrace{T^{BS} + U^{BS}}_{E^{BS}} + T_2$$

3.5 DREHIMPULS IM BS

$$\vec{L}_1^s = m d u (\vec{e}_x - \vec{e}_y) \times (\vec{e}_y - \vec{e}_x) = 0$$

$$\vec{L}_2^s = m d u (-\vec{e}_x - \vec{e}_y) \times (\vec{e}_y - \vec{e}_x) = m d u (-2\vec{e}_z)$$

$$\vec{L}_3^s = m d u 2 \vec{e}_y \times (\vec{e}_x - \vec{e}_y) = m d u (-2\vec{e}_z)$$

$$\vec{L}^s = -4 \vec{e}_z m d u$$

$$\vec{L}_2 = 4 m \vec{R} \times \dot{\vec{R}} = 4 m u d \vec{e}_y \times \vec{e}_x = -4 \vec{e}_z m u d$$

$$\vec{L}^s + \vec{L}_2 = -8 \vec{e}_z m d u = \vec{L} \quad \checkmark$$

Problem 3.6 - Übungsblatt 3

A body on Earth is thrown with the initial velocity $\vec{v}_0 = (v_{0x}, 0, v_{0z})$ from the origin $\vec{r} = \vec{0}$, where \hat{e}_x points towards the north and \hat{e}_z towards the top.

With no apparent forces this body will move along a parabola in the xz -plane.

Determine the deviation of this body from the xz -plane at the end of the flight.

Note: the centrifugal force should be neglected.

The equations of motion therefore contain the Coriolis force only,

$$\text{i.e. } \vec{v}' = -2\vec{\omega} \times \vec{v} + \vec{g}. \quad (*)$$

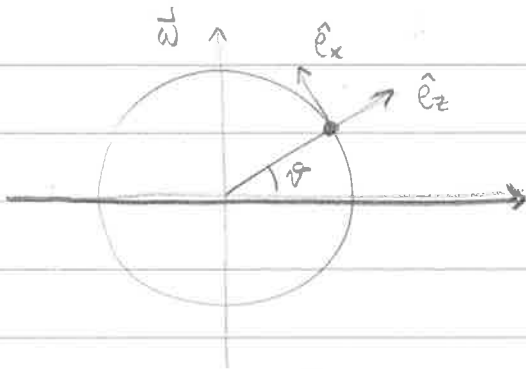
The components of the velocity *in x and z direction* can be determined from $\omega \rightarrow 0$ in equation (*).

Interpret qualitatively the sign of the deflection (east/west) from the point of view of an inertial reference frame.

Solution

$$\vec{v}_0 = (v_{0x}, 0, v_{0z}) \quad \vec{g} = (0, 0, -g)$$

$$\vec{\omega} = (\omega_x, 0, \omega_z) = \omega (\cos \vartheta, 0, \sin \vartheta)$$



the equations of motion read

$$\vec{v}' = \vec{g} - 2\vec{\omega} \times \vec{v} \quad (*)$$

For x, z we can take $\omega \rightarrow 0$ and solve the unperturbed motion, i.e.

$$v_x' = 0 \Rightarrow v_x(t) = v_{0x}$$

$$v_z' = -g \Rightarrow v_z(t) = v_{0z} - gt$$

Taking the y -component of $(*)$ returns

$$v_y' = -2(\vec{\omega} \times \vec{v})_y = -2(\omega_z v_x - \omega_x v_z)$$

$$\vec{\omega} \times \vec{v} = \begin{vmatrix} \hat{i} & \hat{j} & \hat{k} \\ \omega_x & 0 & \omega_z \\ -v_x & v_y & v_z \end{vmatrix} = \begin{pmatrix} -\omega_z v_y \\ \omega_z v_x - \omega_x v_z \\ \omega_x v_y \end{pmatrix}$$

$$v_y' = -2(\omega_z v_x(t) - \omega_x v_z(t)) =$$

$$v_x(t) = v_{0x}$$

$$v_z(t) = v_{0z} - gt$$

$$\dot{v}_y = - \underbrace{2(\omega_z v_{0x} - \omega_x v_{0z})}_{\text{constant}} - 2\omega_x g t$$

$$v_y(t) = - 2(\omega_z v_{0x} - \omega_x v_{0z})t - \frac{2\omega_x g t^2}{2}$$

↑ initial condition $v_{0y} = 0$ (see text)

$$\vec{r} = (x, y, z) \equiv \vec{r}(t)$$

$$x(t) = \int_0^t dt' v_x(t') = v_{0x} t$$

$$y(t) = \int_0^t dt' v_y(t') = -(\omega_z v_{0x} - \omega_x v_{0z})t^2 - \omega_x g \frac{t^3}{3}$$

$$z(t) = \int_0^t dt' v_z(t') = v_{0z} t - \frac{g t^2}{2}$$

with the initial condition $\vec{r}_0 = \vec{r}(0) = (0, 0, 0)$

Flight-time, T_F

$z(t=T_F) \equiv 0$ to find where it lands

$$0 = v_{0z} T_F - \frac{g}{2} T_F^2 \Rightarrow -\frac{2v_{0z}}{g} T_F + T_F^2 = 0$$

$$T_F \left(T_F - \frac{2v_{0z}}{g} \right) = 0$$

$T_F = 0$ to be discarded

$$T_F = \frac{2v_{0z}}{g}$$

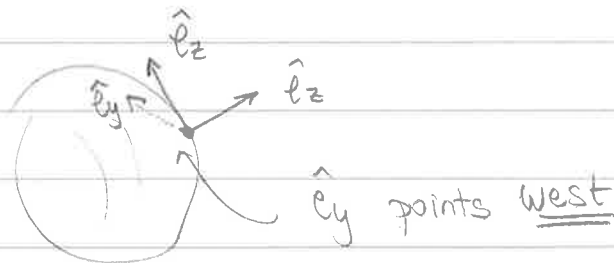
$$\begin{aligned}
 y(t=T_F) &= -(w_z \sqrt{v_{0x}} - w_x \sqrt{v_{0z}}) T_F^2 - \frac{w_x g}{3} T_F^3 = \\
 &= -(w_z \sqrt{v_{0x}} - w_x \sqrt{v_{0z}}) \frac{4 \sqrt{v_{0z}^2}}{g^2} - \frac{w_x g}{3} \left(\frac{8 \sqrt{v_{0z}^3}}{g^3} \right) = \\
 &= \left(\frac{2 \sqrt{v_{0z}}}{g} \right)^2 \left[\underbrace{\frac{\sqrt{v_{0z}} w_x}{3}}_{A \equiv \text{west}} - \underbrace{\sqrt{v_{0x}} w_z}_{\text{east} \equiv B} \right]
 \end{aligned}$$

$w_z > 0$ for the upper hemisphere
 $w_z < 0$ for the lower hemisphere

\hat{e}_y is in direction "west" since

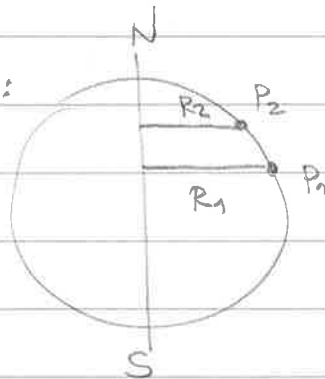
$$\hat{e}_x \times \hat{e}_z = \begin{vmatrix} \hat{i} & \hat{j} & \hat{k} \\ 1 & 0 & 0 \\ 0 & 0 & 1 \end{vmatrix} = \begin{pmatrix} 0 \\ -1 \\ 0 \end{pmatrix} = -\hat{e}_y$$

↑ given the info we have from the text



Interpretation

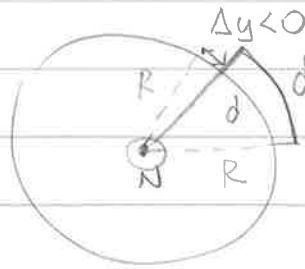
Contribution B:



the mass starts in P_2 ,
 with a velocity $v_{P_2,y} = \omega R_1$
 then lands in P_2 with $R_2 < R_1$
 so $v_{P_2,y} = \omega R_2 < v_{P_1,y}$

Deviation is towards EAST

Contrib. A :



Earth seen from the top
(above the NORTH pole)

$$d = \omega R T_F$$

Deviation towards WEST

